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Pseudospheric shells in the construction

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Abstract. The architects working with the shell use well-established geometry forms, which make up about 5—10 % of the number of known surfaces, in their projects. However, there is such a well-known surface of rotation, which from the 19th century to the present is very popular among mathematicians-geometers, but it is practically unknown to architects and designers, there are no examples of its use in the construction industry. This is a pseudosphere surface. For a pseudospheric surface with a pseudosphere rib radius, the Gaussian curvature at all points equals the constant negative number. The pseudosphere, or the surface of the Beltrami, is generated by the rotation of the tractrix, evolvent of the chain line. The article provides an overview of known methods of calculation of pseudospheric shells and explores the strain-stress state of thin shells of revolution with close geometry parameters to identify optimal forms. As noted earlier, no examples of the use of the surface of the pseudosphere in the construction industry have been found in the scientific and technical literature. Only Kenneth Becher presented examples of pseudospheres implemented in nature: a gypsum model of the pseudosphere made by V. Martin Schilling at the end of the 19th century.

Псевдосферические оболочки в строительстве

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Аннотация. Архитекторы, работающие с оболочкой, используют в своих проектах хорошо зарекомендовавшие себя геометрические формы, которые составляют около 5-10% от числа известных поверхностей. Однако есть такая известная поверхность вращения, которая с XIX в. по настоящее время пользуется большой популярностью среди математиков-геометров, но практически неизвестна

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архитекторам и дизайнерам, нет примеров ее применения в строительной отрасли. Это поверхность псевдосферы. Для псевдосферической поверхности гауссова кривизна во всех точках равна постоянному отрицательному числу. Псевдосфера, или поверхность Бельтрами, образуется вращением трактрисы. Псевдосфера, или поверхность Бельтрами, образуется вращением трассерсиса, эволюционирующего из цепной линии. В статье дается обзор известных методов расчета псевдосферических оболочек и исследуется напряженно-деформированное состояние тонких оболочек вращения с близкими геометрическими параметрами для определения оптимальных форм. Как отмечалось ранее, в научно-технической литературе не найдены примеры применения поверхности псевдосферы в строительной отрасли. Только Кеннет Бехер представил примеры псевдосфер, реализованных в природе: гипсовая модель псевдосферы, сделанная В. Мартином Шиллингом в конце XIX века.

Ключевые слова: псевдосфера, поверхность Бельтрами, трактриса, теория расчета на изгиб, теория временных расчетов, прочность псевдосферических оболочек

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Introduction

The most famous modern architects can be divided into three groups. One group seeks to build [1] high — rise facilities while saving expensive land, especially in urban areas. These architects believe that big — flying structures are necessary only if they are functionally needed. In some cases, large — flying structures have even been demolished to make room for high — rise buildings (Moscow, Minsk, Belarus) or for a more profitable building (The King Dome, Seattle, USA). Another group believes that it is more comfortable for a person to be closer to the ground [2]. They join the former President of the International Shell Association (1966), Prof. A.M. Haas: “The people who build the shells are advanced people; They are united by the desire for new forms, new ways of solving problems” [3] and, by applying scientific approaches to analyse and design large — span structures, they have achieved outstanding success [4]. The third group believes that using traditional constructions (walls, columns, flat overlays) and rectangular shapes, it is also possible to get good results. They focus on low — cost model designs, shapes and high — end materials.

The architects working with the shell use well established geometry forms, which make up about 5—10 % of the number of known surfaces, in their projects. These are paraboloids of revolution [5], umbrella — shaped, apple — shaped, cyclic, propeller, cylindrical shell [6], conic shell [7], mid — surface shell in the form of conoid and cylindroid [8], elliptical paraboloid [9], single —

striped hyperboloid [10] and some others [11]. In each particular case, architects and design engineers chose the most optimal shape of the shell based on functional necessity, strength, aesthetics, etc.

However, there is such a well — known surface of revolution, which from the 19th century to the present is the focus of mathematicians — geometers [12—18], but it is almost unknown to architects and designers, there are no examples of its application in the construction industry. This is a pseudo — sphere surface.

1. Characteristics and methods of setting the pseudosphere

For pseudosphere surface (Figure 1, Figure 2) radius a Gaussian curvature $K = k_1 k_2$ at all points equals a constant negative number $K = -1 / a^2$.

Pseudosphere, or the surface of the Beltram, is formed by rotation of tracersis, chain line involutes $r = ach(z/a)$, relative to the z-axis.

The equation of the tractrix is

$$x = a \sin u, z = a [\cos u + \ln \operatorname{tg}(u/2)], 0 < u < \pi,$$

where u — involute of a catenary line, the angle between the y-axis and tangent to tractices.

The tractrix equation can also be written as

$$z = a \ln \frac{a \pm \sqrt{a^2 - r^2}}{r} \mp \sqrt{a^2 - r^2},$$

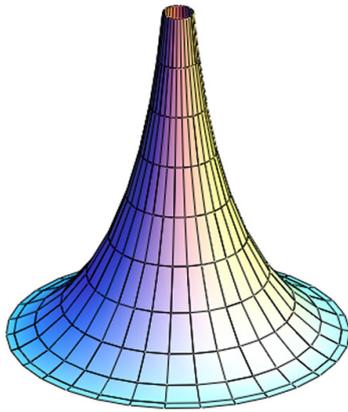


Figure 1. One pseudosphere cavity

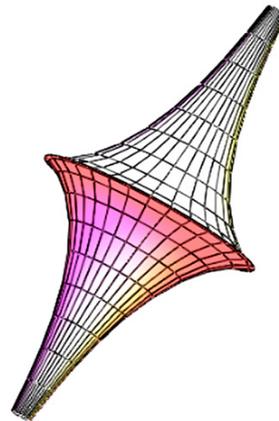


Figure 2. Pseudo-space with two cavities

where the upper signs refer to the positive branch $z > 0$, lower to negative $z < 0$ (Figure 2). The length of the segment tangent to the tractor from the point of contact to the point of intersection with the z -axis – constant and equal to $a > 0$. The section line of the pseudosphere by the xOy plane (edge of the pseudosphere) – is a circle of radius a , for all other parallels $r < a$. The volume of one floor of the pseudosphere: $V = \pi a^3/3$.

Three forms of pseudosphere definition are known using parametric equations [19]. For further application, let us use the following parametric form of pseudosphere surface setting:

$$x = x(r, \beta) = r \cos \beta, \quad y = y(r, \beta) = r \sin \beta,$$

$$z = z(r) = a \ln \left[\left(a + \sqrt{a^2 - r^2} \right) / r \right] - \sqrt{a^2 - r^2},$$

where r is the distance from the rotation axis to the corresponding pseudosphere point ($r < a$), circle $r = r_{max} = a$ – is the pseudosphere edge. Area between parallels $r = a$ and $r = r_0$:

$$S = 2\pi a(a - r_0).$$

In this case, the coefficients of the basic quadratic forms of the surface and its main curvatures are:

$$A = \frac{a}{r}, \quad F = 0, \quad B = r, \quad L = \frac{a}{r\sqrt{a^2 - r^2}},$$

$$M = 0, \quad N = -\frac{r\sqrt{a^2 - r^2}}{a}, \quad k_1 = \frac{r}{a\sqrt{a^2 - r^2}},$$

$$k_2 = -\frac{\sqrt{a^2 - r^2}}{ar}.$$

Therefore, the pseudosphere is defined in curved orthogonal contiguous coordinates. r, β , i.e., in the lines of the main curves. “Appeal in the middle of the XIX century. Geometers to pseudospheric surfaces, surfaces of constant negative curvature $K = -1/a^2$, it was an important step in the development of mathematics. Pseudospheric surfaces were of great importance for the visual interpretation of non-Euclidean hyperbolic geometry discovered by N.I. Lobachevsky. The subsequent development of mathematics revealed a close connection of pseudospheric surfaces with network theory, soliton theory, attractors, nonlinear equations of mathematical physics, Beclund transformations, etc.” [12].

2. Overview of pseudosphere shell analyses

The membrane theory of analysing pseudospheric shells was realized by V.G. Rekach [20] for the case of a homogeneous problem. The solution was made in the form of a trigonometric series. He also determined the tangential forces in the pseudospheric shell of constant thickness from its self-weight, supported hinged-movably in the normal direction in a parallel circle $r = 0,5a$.

The bending theory of analysis of pseudospherical shells in a linear formulation subject to surface symmetric and inversely symmetric loads was considered in the work of D. Werner [21]. The solution was made in analytical form. A numerical example of the analysis is presented in tabular form. To simplify the analysis, the Poisson's ratio was assumed to be zero.

The first attempt to make an overview of all the works devoted to pseudospherical surfaces and shells was made in 1998 in the article [22]. In it, in addition to the investigations of E. Beltrami, V.G. Rekach [20], D. Werner [21], the results obtained by B. Bhattacharya [23] and A.P. Filin [24] are described. A.P. Filin gave formulas for analysing the deformation parameters, the equation of continuity of deformations and the equation of equilibrium of the element of the pseudospherical shell given in the curvature lines. Krawczyk J. [32] considered infinitesimal deformations of thin elastic shells of constant thickness.

A.A. Kalashnikov [30] determined the normal forces in the pseudospherical shell subject to its self-weight by the membrane theory. He then, using the SCAD FEM program, analysed the same shell also subject to its self-weight. Comparison of the results showed a large difference in the values of the ring normal forces on the support. Bending moments are mainly concentrated near the bottom support by the type of edge effect.

In recent years, appeared the first studies on stability of pseudospherical shells. Mikheev A.V. [25], Jasion P., Magnucki K. [26; 27] are working on this issue.

3. Stress-strain state of shells of revolution with close geometric parameters

There are a number of works, for example [29], where some criteria are put forward for assessing the optimality of the selected design solution. V.V. Novozhilov [29] suggested using the results of their analysis according to the membrane theory for an approximate estimate of the optimality of the chosen form of the thin-walled shell of revolution. Shells with similar geometric parameters (boom lifting and diameter of the shell at the base) were chosen for the analysis.

Let's follow his example. Figure 3 shows five types of shells of revolution. The pseudospherical (Figure 3, a),

conical shell (Figure 3, b), a shell with a median surface of revolution of the hyperbola $z = \frac{b}{x}$ around the z-axis (Figure 3, c), a shell with the median surface of revolution of the asteroid (Figure 3, d) and in the form of a one-sheeted hyperboloid of revolution (Figure 3, e).

All these surfaces are defined by parametric equations: $x = x(r, \beta) = r \cos \beta$, $y = y(r, \beta) = r \sin \beta$, $z = z(r)$, where for the pseudosphere (Figure 3, a):

$$z = z(r) = a \ln \frac{a + \sqrt{a^2 - r^2}}{r} - \sqrt{a^2 - r^2},$$

for the cone (Figure 3, b):

$$z = z(r) = -\frac{rH}{r_1 - r_2},$$

for the surface of rotation of the hyperbola $z = b/x$ around the axis Oz (Figure 3):

$$z = z(r) = \frac{Hr_1r_2}{(r_1 - r_2)r}.$$

For the surface of the rotation of the asteroid (Figure 3, g):

$$z = z(r) = (b^{\frac{2}{3}} - r^{\frac{2}{3}})^{3/2},$$

the parameter b must be found from the equality:

$$H = (b^{2/3} - r_2^{2/3})^{3/2} - (b^{2/3} - r_1^{2/3})^{3/2},$$

for a single-cavity hyperboloid of revolution (Figure 3, d):

$$z = z(r) = \frac{-H\sqrt{r^2 - b^2}}{(\sqrt{r_1^2 - b^2} - \sqrt{r_2^2 - b^2})}.$$

where parameter b can take any value, but $b < r_2$.

All unspecified geometric parameters are shown in Figure 3. In the same figure, the meridians of the pseudosphere are shown by a solid line, and the meridians of the remaining surfaces of revolution are shown by a thin line with dots.

Shell thickness $h = 0.05 \text{ cm}$, self-weight type surface load $q = 100 \text{ kg} / \text{m}^2$, the radius of the base is $r_1 = 4 \text{ m}$, the radius of the hole in the apex is $r_2 = 2 \text{ m}$, the boom of lifting $H = 5.15 \text{ m}$ is the same for all shells, $r_2 \leq r \leq r_1$, $0 \leq \beta \leq 2\pi$.

Under axisymmetric loading of the shells of revolution, the surface distributed load in the direction of the curvilinear coordinate β is zero ($Y = 0$), normal forces (N_r, N_β), shearing forces (Q_r), bending moments (M_r, M_β), deformations ($\varepsilon_r, \varepsilon_\beta, \kappa_r, \kappa_\beta$) and displacements

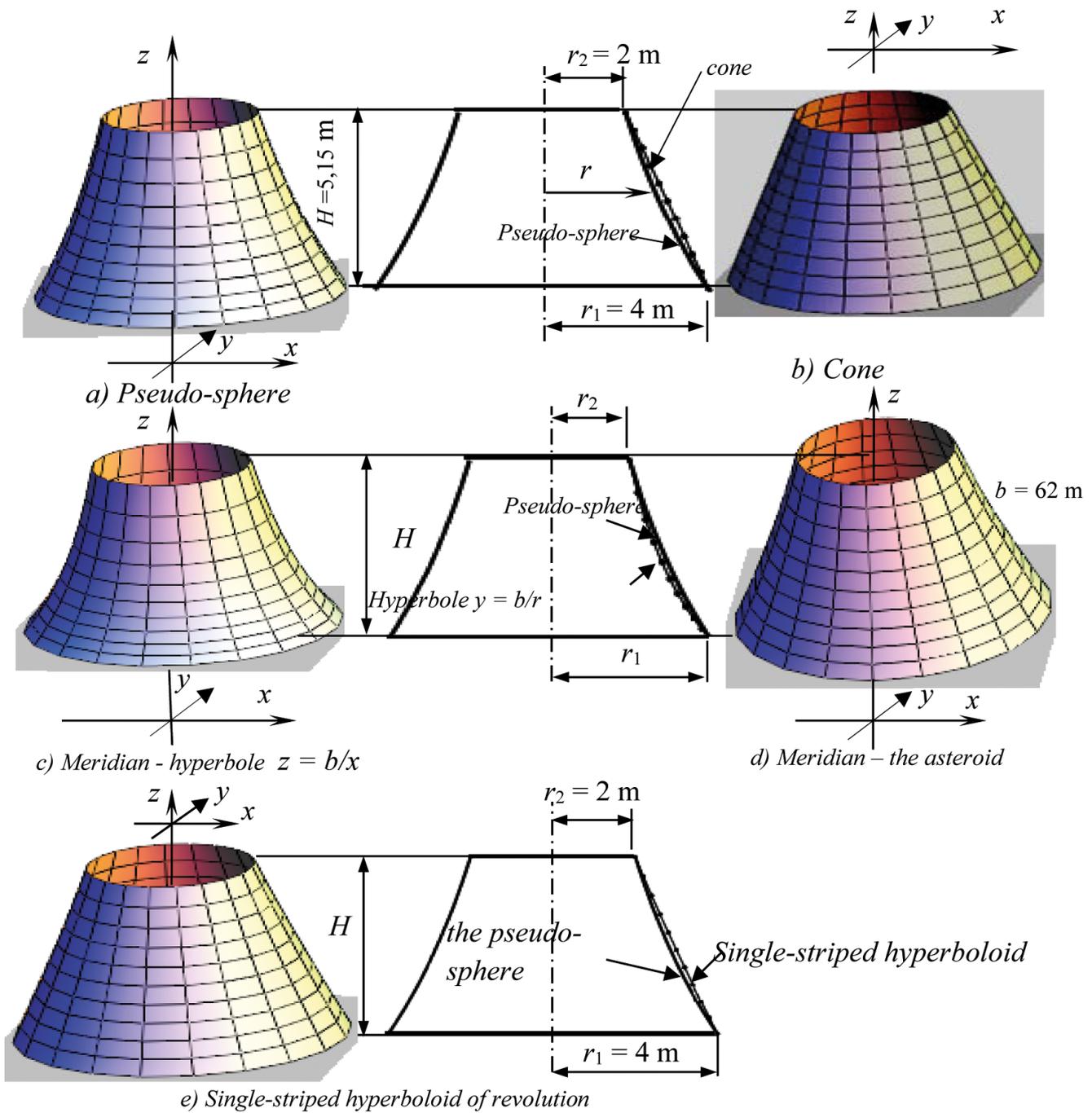


Figure 3. Five types of rotation surfaces

$(W = u_z, u_r)$ are independent of the longitude angle β , and, in addition,

$$S = Q_\beta = M_{r\beta} = 0, \quad u_\beta = \varepsilon_{r\beta} = \kappa_{r\beta} = 0.$$

Figures 4–28 present the analysis results of the shells of revolution considered subject to self-weight by the finite elements method with a pivotally fixed support of

the lower edge ($r = r_1$) and the free upper edge ($r = r_2$). Accepted $E = 3.5 \cdot 10^4\text{ MPa}$, Poisson's ratio $\nu = 0.1$.

	M	M	
✓	0	1.39e-006	300
✓	1.39e-006	2.77e-006	200
✓	2.77e-006	4.16e-006	100
✓	4.16e-006	5.54e-006	200
✓	5.54e-006	6.93e-006	200
✓	6.93e-006	8.31e-006	200
✓	8.31e-006	9.7e-006	100
✓	9.7e-006	1.11e-005	200
✓	1.11e-005	1.25e-005	200
✓	1.25e-005	1.39e-005	1000
✓	1.39e-005	1.52e-005	200
✓	1.52e-005	1.66e-005	1000
✓	1.66e-005	1.8e-005	1500
✓	1.8e-005	1.94e-005	1000
✓	1.94e-005	2.08e-005	500
✓	2.08e-005	2.22e-005	800

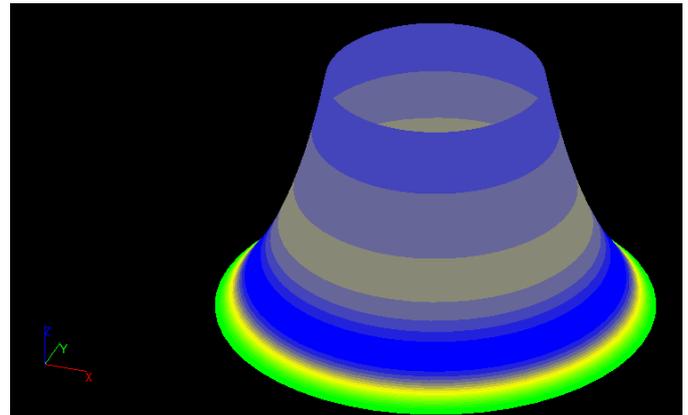


Figure 4. The overall displacement of Pseudosphere

	M	M	
✓	0	8.31e-007	200
✓	8.31e-007	1.66e-006	300
✓	1.66e-006	2.49e-006	200
✓	2.49e-006	3.33e-006	200
✓	3.33e-006	4.16e-006	200
✓	4.16e-006	4.99e-006	300
✓	4.99e-006	5.82e-006	300
✓	5.82e-006	6.65e-006	300
✓	6.65e-006	7.48e-006	300
✓	7.48e-006	8.31e-006	400
✓	8.31e-006	9.15e-006	400
✓	9.15e-006	9.98e-006	400
✓	9.98e-006	1.08e-005	500
✓	1.08e-005	1.16e-005	500
✓	1.16e-005	1.25e-005	600
✓	1.25e-005	1.33e-005	1400

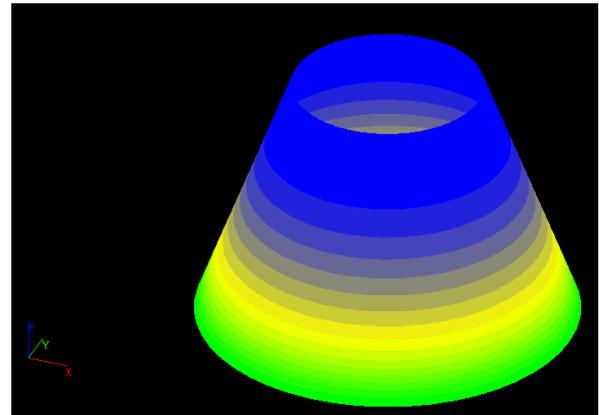


Figure 5. The overall displacement of Cone

	M	M	
✓	0	9.94e-007	200
✓	9.94e-007	1.99e-006	200
✓	1.99e-006	2.98e-006	200
✓	2.98e-006	3.98e-006	200
✓	3.98e-006	4.97e-006	200
✓	4.97e-006	5.96e-006	100
✓	5.96e-006	6.96e-006	200
✓	6.96e-006	7.95e-006	200
✓	7.95e-006	8.95e-006	100
✓	8.95e-006	9.94e-006	200
✓	9.94e-006	1.09e-005	200
✓	1.09e-005	1.19e-005	200
✓	1.19e-005	1.29e-005	200
✓	1.29e-005	1.39e-005	1700
✓	1.39e-005	1.49e-005	1600
✓	1.49e-005	1.59e-005	1000

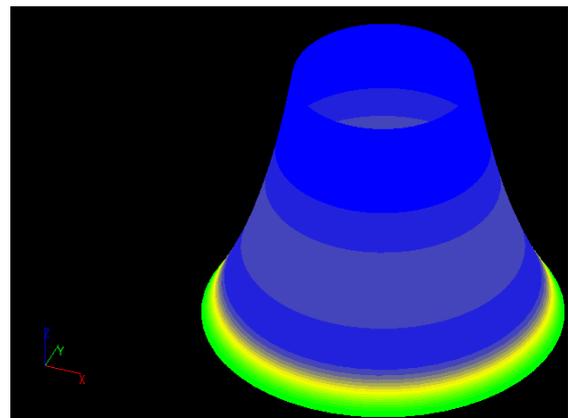


Figure 6. The overall displacement of Meridian – Hyperbole

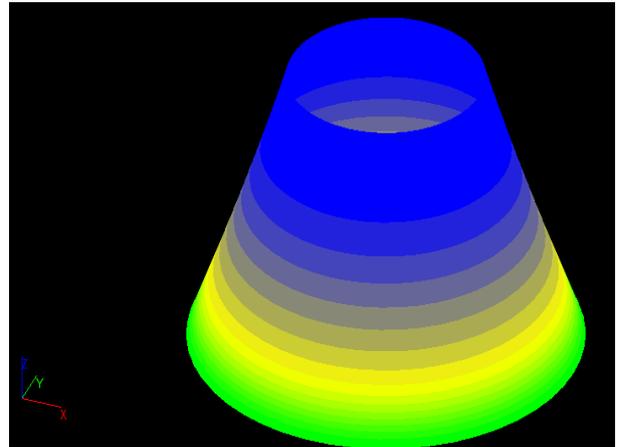
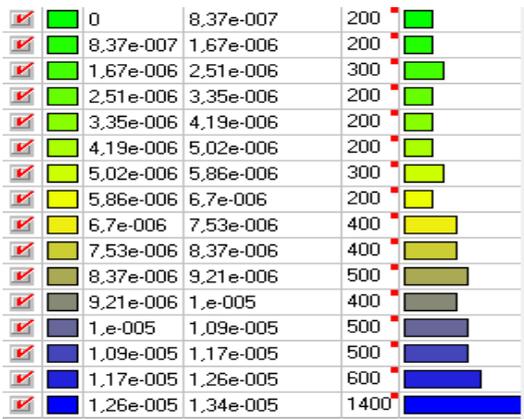


Figure 7. The overall displacement of Meridian – Asteroid

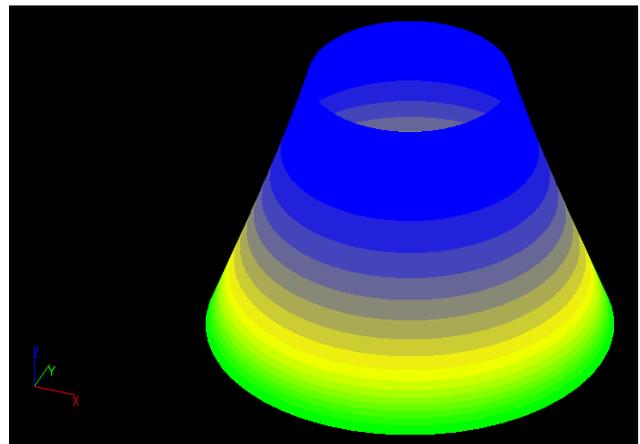
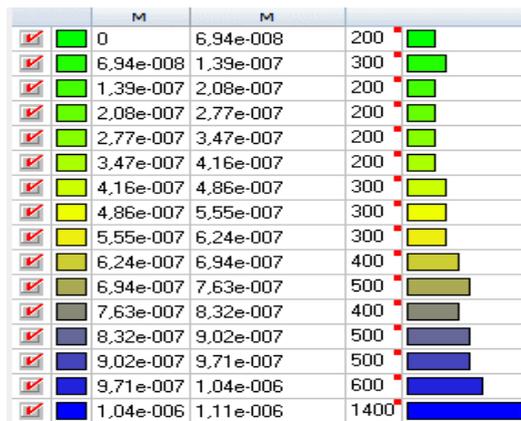


Figure 8. The overall displacement of Single-cavity hyperboloid of revolution

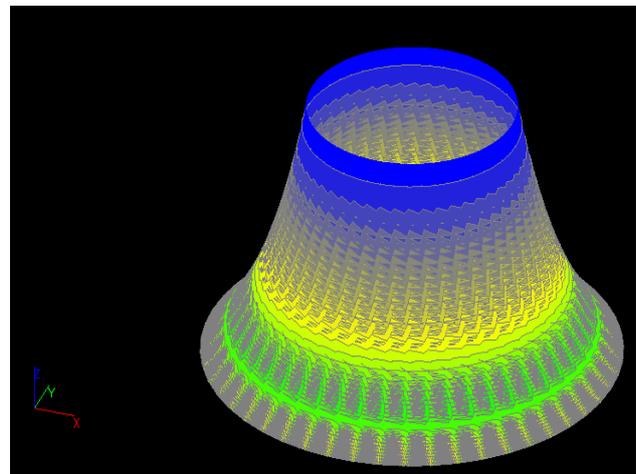
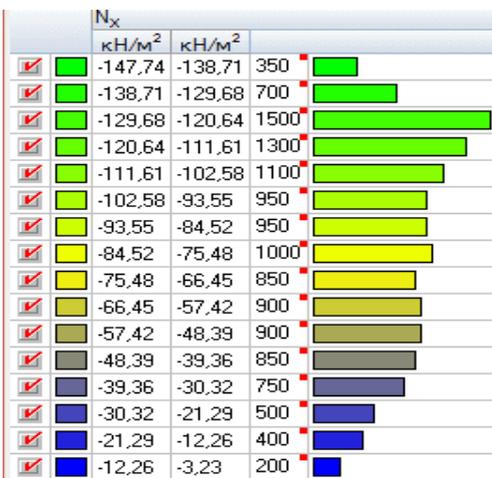


Figure 9. The Normal effort of Pseudosphere about the x-axis

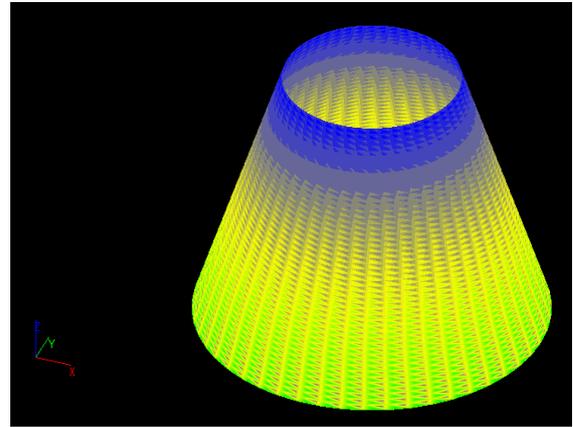
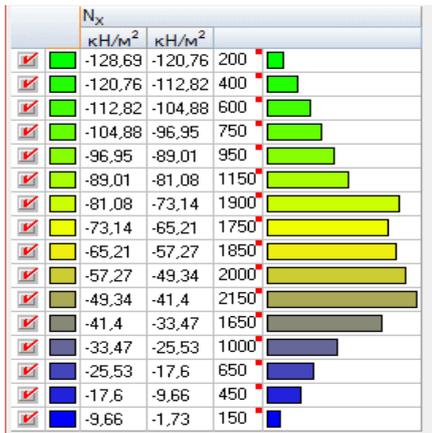


Figure 10. The Normal effort of Cone about the x-axis

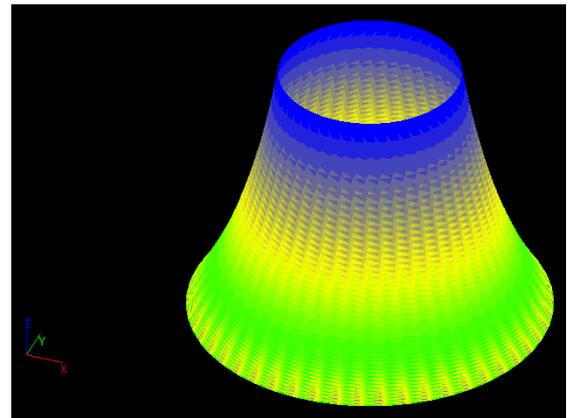
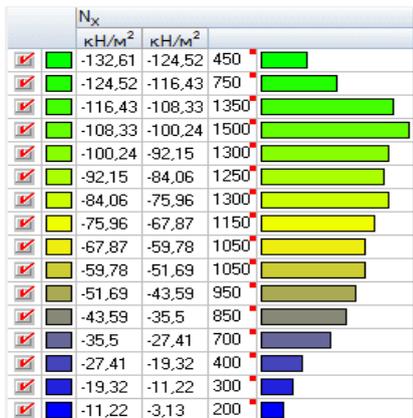


Figure 11. The Normal effort of Meridian – Hyperbole about the x-axis

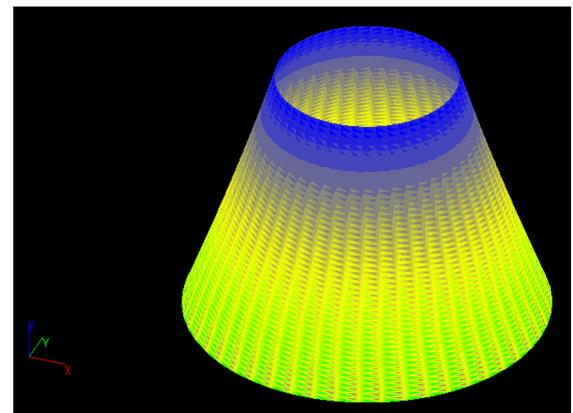
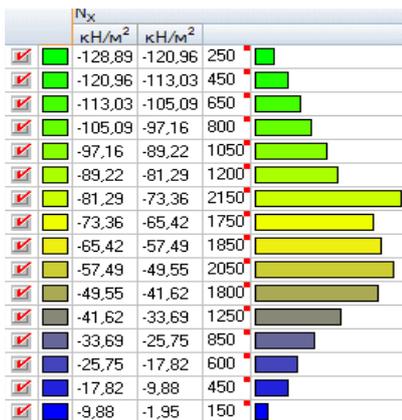


Figure 12. The Normal effort of Meridian – Asteroid about the x-axis

		N_x		
		$\kappa\text{H}/\text{M}^2$	$\kappa\text{H}/\text{M}^2$	
✓	█	-10,7	-10,04	250
✓	█	-10,04	-9,38	450
✓	█	-9,38	-8,73	650
✓	█	-8,73	-8,07	800
✓	█	-8,07	-7,41	1050
✓	█	-7,41	-6,75	1200
✓	█	-6,75	-6,1	2100
✓	█	-6,1	-5,44	1850
✓	█	-5,44	-4,78	1950
✓	█	-4,78	-4,12	2000
✓	█	-4,12	-3,47	2050
✓	█	-3,47	-2,81	1350
✓	█	-2,81	-2,15	750
✓	█	-2,15	-1,49	550
✓	█	-1,49	-0,83	450
✓	█	-0,83	-0,18	200

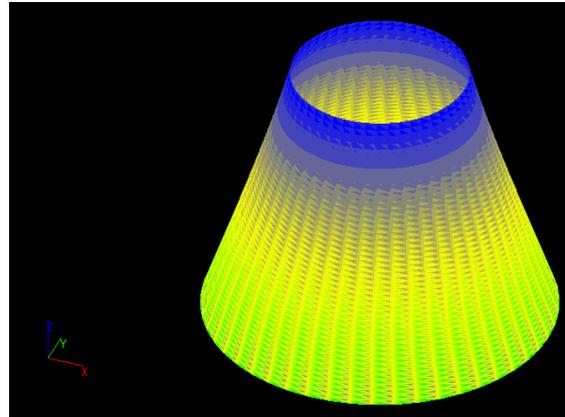


Figure 13. The Normal effort of Single-cavity hyperboloid of revolution about the x-axis

		N_y		
		$\kappa\text{H}/\text{M}^2$	$\kappa\text{H}/\text{M}^2$	
✓	█	-148,5	-139,61	300
✓	█	-139,61	-130,71	1250
✓	█	-130,71	-121,82	1500
✓	█	-121,82	-112,92	1350
✓	█	-112,92	-104,02	1150
✓	█	-104,02	-95,13	1050
✓	█	-95,13	-86,23	900
✓	█	-86,23	-77,33	950
✓	█	-77,33	-68,44	850
✓	█	-68,44	-59,54	850
✓	█	-59,54	-50,64	850
✓	█	-50,64	-41,75	850
✓	█	-41,75	-32,85	800
✓	█	-32,85	-23,96	700
✓	█	-23,96	-15,06	500
✓	█	-15,06	-6,16	300

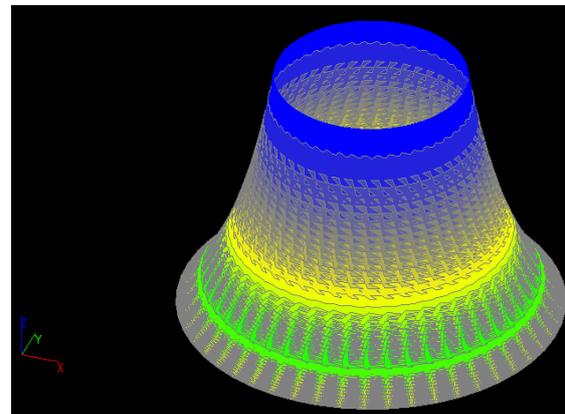


Figure 14. The Normal effort of Pseudosphere about the y-axis

		N_y		
		$\kappa\text{H}/\text{M}^2$	$\kappa\text{H}/\text{M}^2$	
✓	█	-123,74	-116,34	250
✓	█	-116,34	-108,93	400
✓	█	-108,93	-101,52	600
✓	█	-101,52	-94,11	750
✓	█	-94,11	-86,7	900
✓	█	-86,7	-79,3	1050
✓	█	-79,3	-71,89	1750
✓	█	-71,89	-64,48	1750
✓	█	-64,48	-57,07	1850
✓	█	-57,07	-49,66	1950
✓	█	-49,66	-42,26	2100
✓	█	-42,26	-34,85	2250
✓	█	-34,85	-27,44	1550
✓	█	-27,44	-20,03	900
✓	█	-20,03	-12,62	550
✓	█	-12,62	-5,22	200

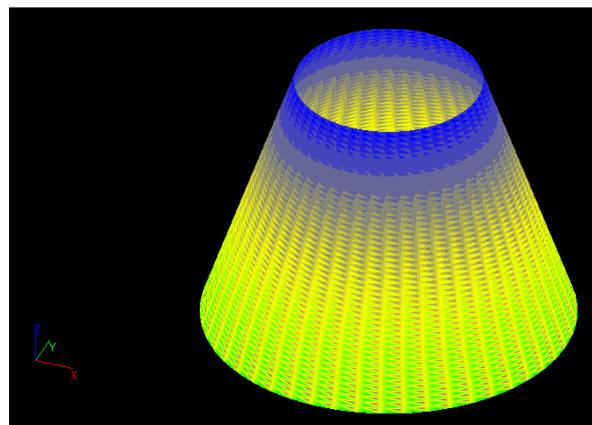


Figure 15. The Normal effort of Cone about the y-axis

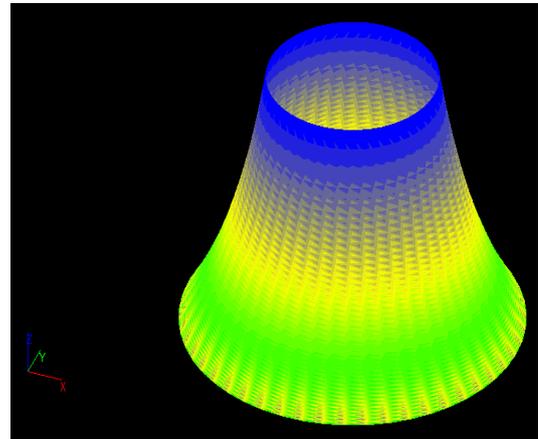
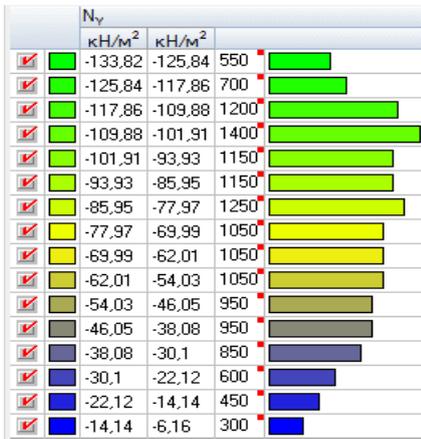


Figure 16. The Normal effort of Meridian – Hyperbole about the x-axis

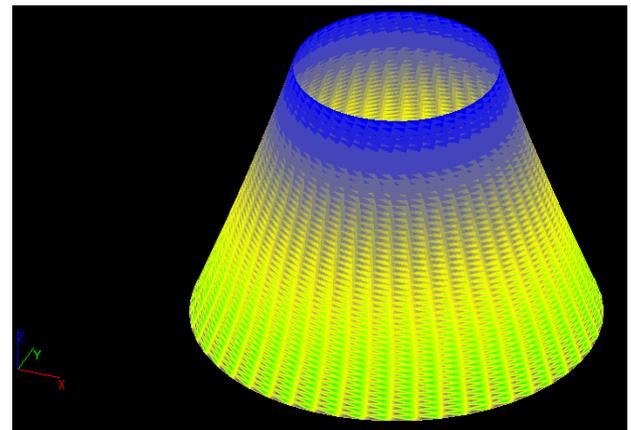


Figure 17. The Normal effort of Meridian – Asteroid about the y-axis

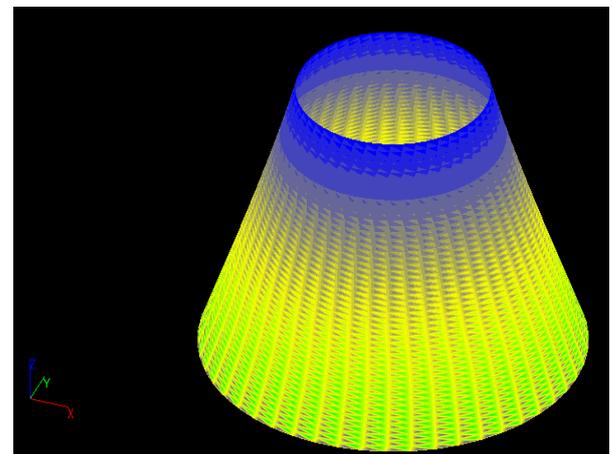
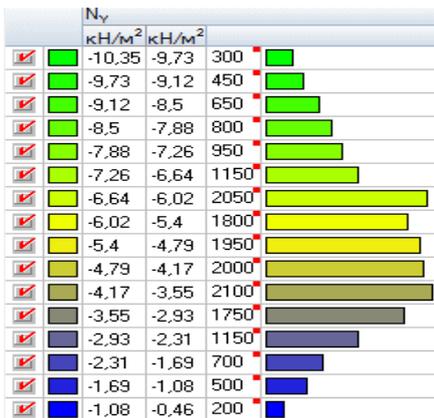


Figure 18. The Normal effort of Single-cavity hyperboloid of revolution about the y-axis

		M_x			
		кН*м/м	кН*м/м		
<input checked="" type="checkbox"/>	█	-0,6	-0,54	50	█
<input checked="" type="checkbox"/>	█	-0,54	-0,46	50	█
<input checked="" type="checkbox"/>	█	-0,46	-0,38	100	█
<input checked="" type="checkbox"/>	█	-0,38	-0,31	200	█
<input checked="" type="checkbox"/>	█	-0,31	-0,23	300	█
<input checked="" type="checkbox"/>	█	-0,23	-0,15	300	█
<input checked="" type="checkbox"/>	█	-0,15	-0,08	350	█
<input checked="" type="checkbox"/>	█	-0,08	0	3000	█
<input checked="" type="checkbox"/>	█	0	0,08	1700	█
<input checked="" type="checkbox"/>	█	0,08	0,15	1000	█
<input checked="" type="checkbox"/>	█	0,15	0,17	200	█

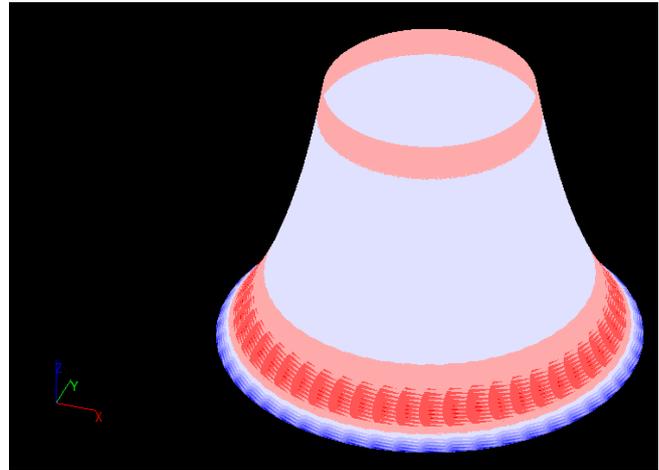


Figure 19. The bending Moment of Pseudosphere about the x-axis

		M_x			
		кН*м/м	кН*м/м		
<input checked="" type="checkbox"/>	█	-0,06	-0,06	50	█
<input checked="" type="checkbox"/>	█	-0,06	-0,05	50	█
<input checked="" type="checkbox"/>	█	-0,05	-0,04	100	█
<input checked="" type="checkbox"/>	█	-0,04	-0,04	100	█
<input checked="" type="checkbox"/>	█	-0,04	-0,03	150	█
<input checked="" type="checkbox"/>	█	-0,03	-0,02	200	█
<input checked="" type="checkbox"/>	█	-0,02	-0,01	250	█
<input checked="" type="checkbox"/>	█	-0,01	-0,01	200	█
<input checked="" type="checkbox"/>	█	-0,01	0	200	█
<input checked="" type="checkbox"/>	█	0	0,01	4550	█
<input checked="" type="checkbox"/>	█	0,01	0,01	600	█

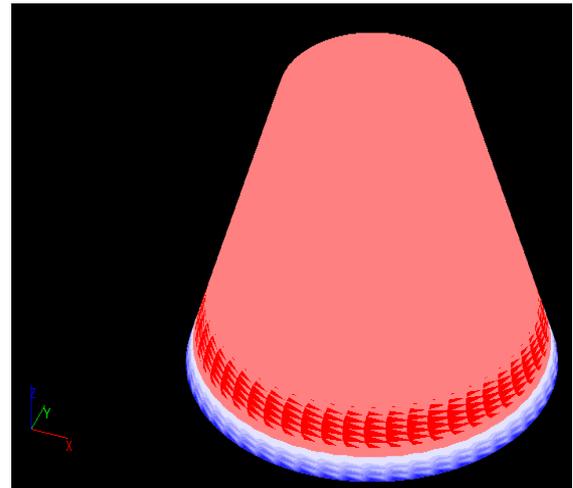


Figure 20. The bending Moment of Cone about the x-axis

		M_x			
		кН*м/м	кН*м/м		
<input checked="" type="checkbox"/>	█	-0,32	-0,29	50	█
<input checked="" type="checkbox"/>	█	-0,29	-0,25	50	█
<input checked="" type="checkbox"/>	█	-0,25	-0,21	150	█
<input checked="" type="checkbox"/>	█	-0,21	-0,17	200	█
<input checked="" type="checkbox"/>	█	-0,17	-0,12	200	█
<input checked="" type="checkbox"/>	█	-0,12	-0,08	300	█
<input checked="" type="checkbox"/>	█	-0,08	-0,04	300	█
<input checked="" type="checkbox"/>	█	-0,04	0	2950	█
<input checked="" type="checkbox"/>	█	0	0,04	1800	█
<input checked="" type="checkbox"/>	█	0,04	0,08	900	█
<input checked="" type="checkbox"/>	█	0,08	0,09	200	█

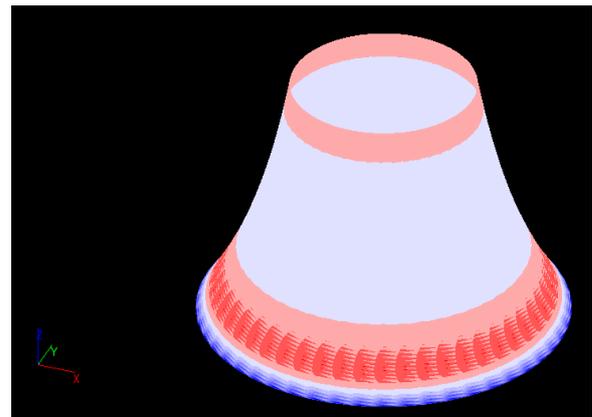


Figure 21. The bending Moment of Meridian – Hyperbole about the x-axis

		M_x		
		$\kappa H^2 m/m$	$\kappa H^2 m/m$	
<input checked="" type="checkbox"/>		-0,09	-0,09	50
<input checked="" type="checkbox"/>		-0,09	-0,08	50
<input checked="" type="checkbox"/>		-0,08	-0,06	100
<input checked="" type="checkbox"/>		-0,06	-0,05	100
<input checked="" type="checkbox"/>		-0,05	-0,04	150
<input checked="" type="checkbox"/>		-0,04	-0,03	200
<input checked="" type="checkbox"/>		-0,03	-0,02	250
<input checked="" type="checkbox"/>		-0,02	-0,01	200
<input checked="" type="checkbox"/>		-0,01	0	950
<input checked="" type="checkbox"/>		0	0,01	4000
<input checked="" type="checkbox"/>		0,01	0,02	700

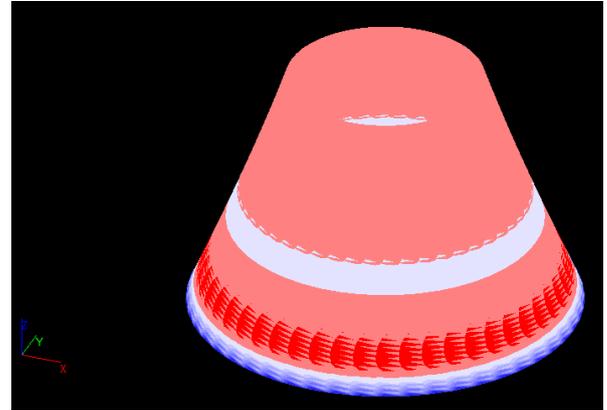


Figure 22. The bending Moment of Meridian – Asteroid about the y-axis

		M_x		
		$\kappa H^2 m/m$	$\kappa H^2 m/m$	
<input checked="" type="checkbox"/>		-0,01	-0,01	50
<input checked="" type="checkbox"/>		-0,01	-0,01	50
<input checked="" type="checkbox"/>		-0,01	-4,77e-003	100
<input checked="" type="checkbox"/>		-4,77e-003	-3,97e-003	100
<input checked="" type="checkbox"/>		-3,97e-003	-3,18e-003	150
<input checked="" type="checkbox"/>		-3,18e-003	-2,38e-003	200
<input checked="" type="checkbox"/>		-2,38e-003	-1,59e-003	250
<input checked="" type="checkbox"/>		-1,59e-003	-7,94e-004	200
<input checked="" type="checkbox"/>		-7,94e-004	0	200
<input checked="" type="checkbox"/>		0	7,94e-004	4500
<input checked="" type="checkbox"/>		7,94e-004	1,46e-003	700

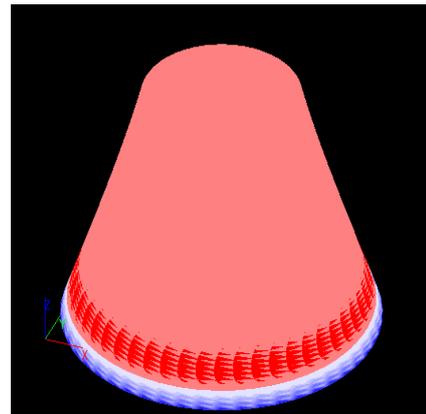


Figure 23. The bending Moment of Single-cavity hyperboloid of revolution about the x-axis

		M_y		
		$\kappa H^2 m/m$	$\kappa H^2 m/m$	
<input checked="" type="checkbox"/>		-0,59	-0,54	50
<input checked="" type="checkbox"/>		-0,54	-0,46	100
<input checked="" type="checkbox"/>		-0,46	-0,38	100
<input checked="" type="checkbox"/>		-0,38	-0,31	150
<input checked="" type="checkbox"/>		-0,31	-0,23	250
<input checked="" type="checkbox"/>		-0,23	-0,15	350
<input checked="" type="checkbox"/>		-0,15	-0,08	400
<input checked="" type="checkbox"/>		-0,08	0	2850
<input checked="" type="checkbox"/>		0	0,08	1900
<input checked="" type="checkbox"/>		0,08	0,15	950
<input checked="" type="checkbox"/>		0,15	0,18	300

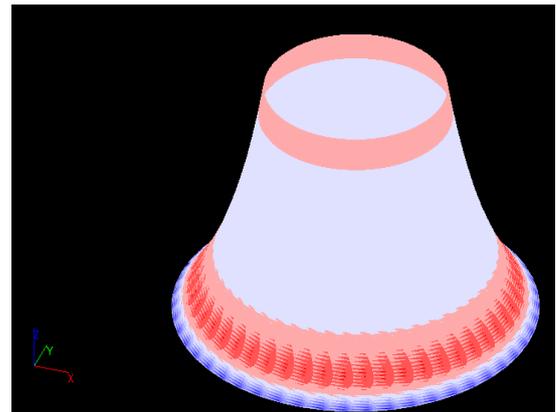


Figure 24. The bending Moment of Pseudosphere about the y-axis

		M_y		
		кН*м/м	кН*м/м	
<input checked="" type="checkbox"/>	■	-0,05	-0,05	50
<input checked="" type="checkbox"/>	■	-0,05	-0,05	50
<input checked="" type="checkbox"/>	■	-0,05	-0,04	50
<input checked="" type="checkbox"/>	■	-0,04	-0,03	50
<input checked="" type="checkbox"/>	■	-0,03	-0,03	200
<input checked="" type="checkbox"/>	■	-0,03	-0,02	200
<input checked="" type="checkbox"/>	■	-0,02	-0,01	250
<input checked="" type="checkbox"/>	■	-0,01	-0,01	200
<input checked="" type="checkbox"/>	■	-0,01	0	250
<input checked="" type="checkbox"/>	■	0	0,01	4550
<input checked="" type="checkbox"/>	■	0,01	0,01	600

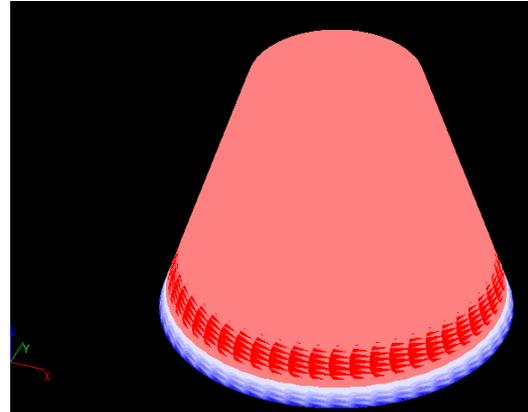


Figure 25. The bending Moment of Cone about the y-axis

		кН*м/м	кН*м/м	
■		-0,32	-0,29	50
■		-0,29	-0,25	50
■		-0,25	-0,2	100
■		-0,2	-0,16	100
■		-0,16	-0,12	250
■		-0,12	-0,08	350
■		-0,08	-0,04	300
■		-0,04	0	3050
■		0	0,04	1850
■		0,04	0,08	950
■		0,08	0,09	250

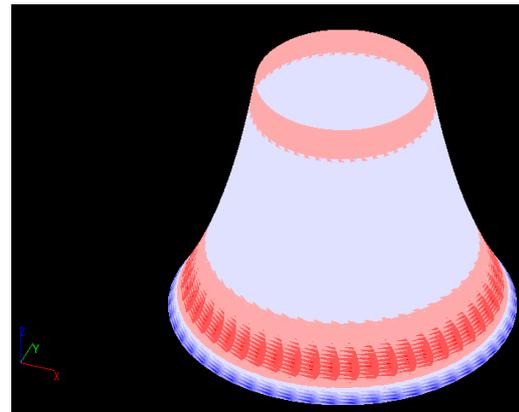


Figure 26. The bending Moment of Meridian – Hyperbole about the y-axis

		M_y		
		кН*м/м	кН*м/м	
<input checked="" type="checkbox"/>	■	-0,08	-0,08	50
<input checked="" type="checkbox"/>	■	-0,08	-0,07	50
<input checked="" type="checkbox"/>	■	-0,07	-0,06	50
<input checked="" type="checkbox"/>	■	-0,06	-0,05	50
<input checked="" type="checkbox"/>	■	-0,05	-0,04	200
<input checked="" type="checkbox"/>	■	-0,04	-0,03	200
<input checked="" type="checkbox"/>	■	-0,03	-0,02	250
<input checked="" type="checkbox"/>	■	-0,02	-0,01	300
<input checked="" type="checkbox"/>	■	-0,01	0	950
<input checked="" type="checkbox"/>	■	0	0,01	4100
<input checked="" type="checkbox"/>	■	0,01	0,02	700

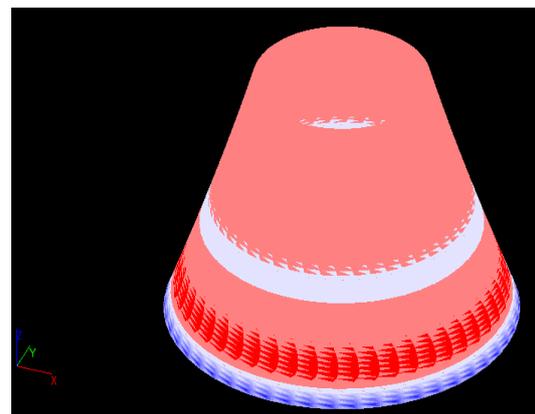


Figure 27. The bending-Moment of Meridian – Asteroid about the y-axis

		M_y		
		$\kappa H^2 M/m$	$\kappa H^2 M/m$	
<input checked="" type="checkbox"/>		-0,01	-0,01	50
<input checked="" type="checkbox"/>		-0,01	-0,01	50
<input checked="" type="checkbox"/>		-0,01	-4,53e-003	50
<input checked="" type="checkbox"/>		-4,53e-003	-3,78e-003	100
<input checked="" type="checkbox"/>		-3,78e-003	-3,02e-003	200
<input checked="" type="checkbox"/>		-3,02e-003	-2,27e-003	200
<input checked="" type="checkbox"/>		-2,27e-003	-1,51e-003	250
<input checked="" type="checkbox"/>		-1,51e-003	-7,55e-004	300
<input checked="" type="checkbox"/>		-7,55e-004	0	300
<input checked="" type="checkbox"/>		0	7,55e-004	4500
<input checked="" type="checkbox"/>		7,55e-004	1,45e-003	650

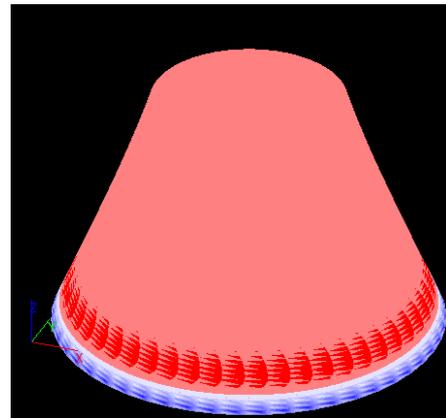


Figure 28. The bending-Moment of Single-cavity hyperboloid of revolution about the y-axis

4. Proposals for the use of pseudospherical shells in architecture and the construction industry



Figure 29. Stainless steel sculpture "Non Object", A. Kapoor, 2008

As noted earlier, no examples of the use of the surface of the pseudosphere in the construction industry have been found in the scientific and technical literature.

Only Kenneth Brecher [18] presented examples of pseudospheres implemented in nature: a gypsum model of the pseudosphere made by V.M. Schilling (V. Martin Schilling) at the end of the XIX century. The large-sized plywood pseudosphere model, *Mathematica*, exhibited at the Boston Science Museum, created by Charles and Ray Eames. Stainless steel sculpture for the park made by Anish Kapoor, 2008, fig. 26). The author — H. Sugimoto, 2004, made the model “The surface of revolution of constant negative curvature” and a sculpture made of aluminum and glass by the same author “Conceptual Form 009”, 2006. Robert Le Ricolais make the “Funicular polygon of revolution — pseudosphere” metal wire shape. All these mathematical models serve educational purposes.

B. Bhattacharya [23] proposed to use a pseudo-spherical shell as the foundations of reinforced concrete chimneys. D. Werner [21] proposed to use a pseudo-spherical thin-walled shell as the base of the tower of the television station.

Basing his research on parametric equations of the pseudosphere, Zh. Kaydasov [31] introduced a new type of surfaces-multi-tube “pseudospheres” with external cycloidal and sinusoidal corrugations. In his opinion, these surfaces can attract the attention of architects. We take the name of the surface — “pseudosphere” — in quotes, because the proposed surface does not have a constant negative Gaussian curvature.

Additional information on the application of pseudospherical shells in building industry can be taken in a manuscript [33].

Conclusion

The areas of fragments of the considered surfaces of revolution with close geometric parameters are naturally compared with the minimum surface of revolution — the catenoid — the area of the fragment of which is determined by the formula:

$$A_{\text{кат}} = \pi \left[a^2 \ln |r + \sqrt{r^2 - a^2}| + r\sqrt{r^2 - a^2} \right]_{r_2}^{r_1},$$

where $a \leq r_2$ — is the radius of the throat circumference lying in the xOy , plane, which is determined from the expression:

$$H = a \left(\text{Arch} \frac{r_1}{a} - \text{Arch} \frac{r_2}{a} \right). \quad (1)$$

The above two formulas must be used if the throat circumference of the catenoid is outside the considered fragment of the catenoid.

If the throat circumference is within the considered surface, as in our case, then these two formulas must be written in the following form:

$$A_{\text{кат}} = \pi \left[a^2 \ln |r + \sqrt{r^2 - a^2}| + r\sqrt{r^2 - a^2} \right]_{r_2}^a + \pi \left[a^2 \ln |r + \sqrt{r^2 - a^2}| + r\sqrt{r^2 - a^2} \right]_{r_1}^a,$$

where $a \leq r_2$ — the radius of the throat circle lying in the xOy , plane, which is defined from the expression:

$$H = a \left(\text{Arch} \frac{r_1}{a} + \text{Arch} \frac{r_2}{a} \right). \quad (2)$$

You can take the optimality factor over the area in the form:

$$\rho_{\text{pow}} = \frac{A_{\text{pow}}}{A_{\text{cat}}} \geq 1.$$

Naturally, for the catenoid in this case $\rho_{\text{pow}} = 1$. However, for the case in question, $r_1 = 4$ m, $r_2 = 2$ m, $H = 5.15$ m the formulas (1), (2) have no solution, so compare the shells in question (Fig. 3) with a shell having a median minimum surface with the same geometric parameters, it is impossible. You can select the inverse task. First determine by the formula (2) $H_{\text{max}} = 3.65$ m at $a = 1.448$ m for $r_1 = 4$ m, $r_2 = 2$ m, and then pick up the other shells of revolution.

We introduce the optimality factor over the area:

$$\rho_{\text{пов}} = \frac{A_{\text{пов}}}{A_{\text{псевд}}}.$$

The analyses showed that of the five shells presented in Fig. 3, the smallest area of the median surface has a shell with a hyperbola $z = b/x$ as a Meridian ($\rho = 98.25/100.5 = 0.98$). It is followed by a pseudosphere ($\rho = 1$), a shell with an asteroid ($\rho = 102.86/100.5 = 1.02$) as a Meridian, a unicellular hyperboloid of revolution ($\rho = 103.1/100.5 = 1.03$) and a cone ($\rho = 104.1/100.5 = 1.04$). The difference in surface areas shown in Fig. 3 is very small.

Comparing the isofields of efforts and displacements presented in Fig. 4—28, we can conclude that the pseudospherical shell has no particular advantages in this indicator either.

V.V. Novozhilov [29] wrote about the lack of demand for the catenoid, i.e. the only minimum surface rotation in construction in the near future. It will probably also be with a pseudospherical shell.

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